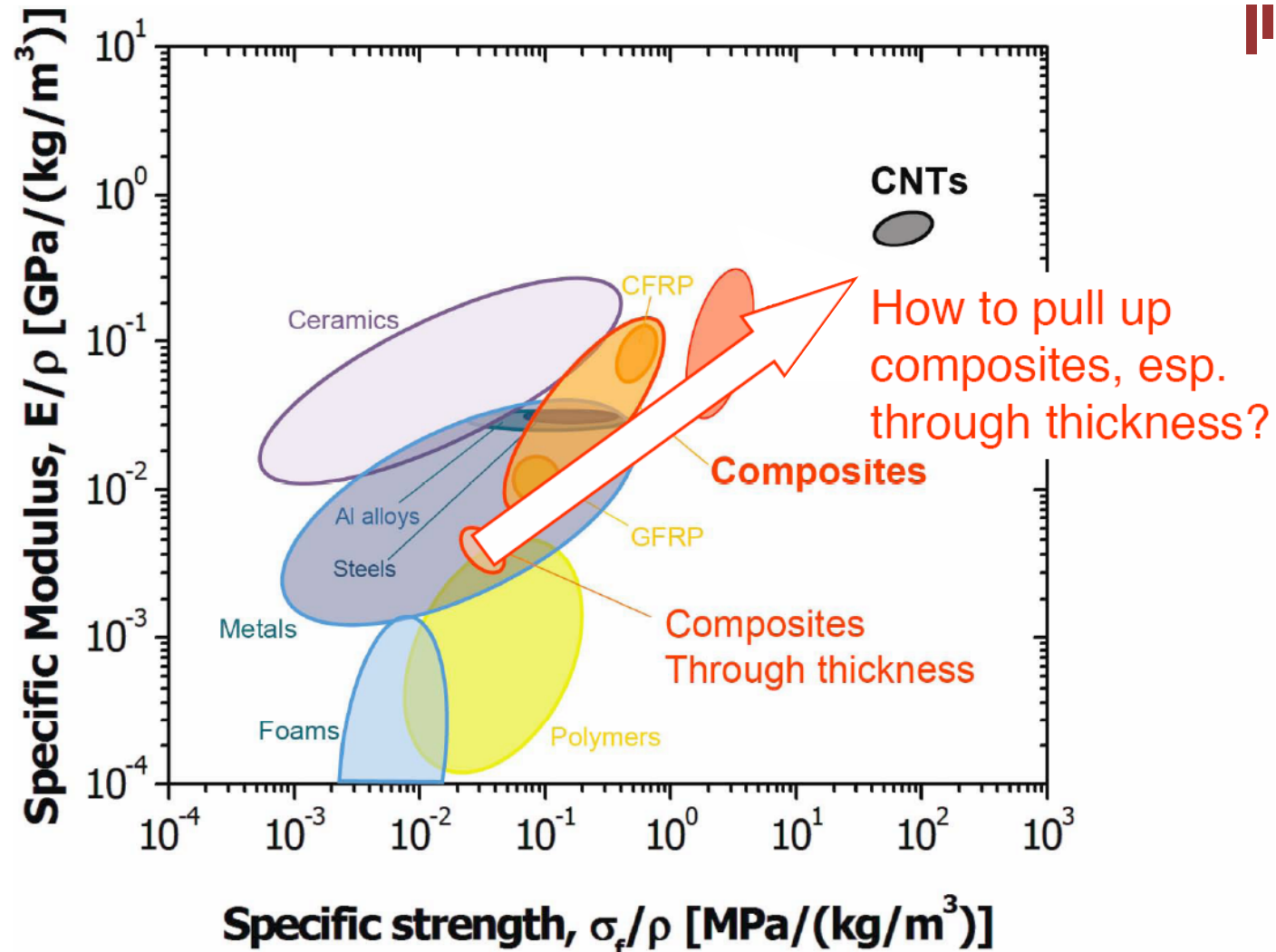


Carbon Nanotube (CNT) Enhancements for --- Aerosurface State Awareness

Dr. Seth S. Kessler & Dr. Christopher Dunn
Metis Design Corporation

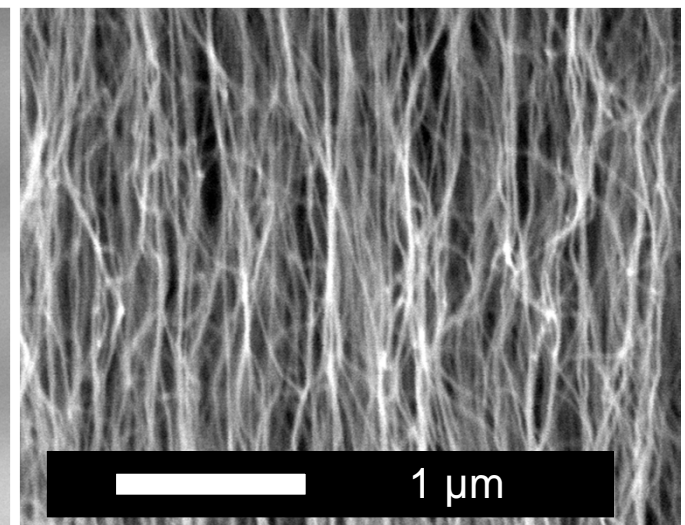
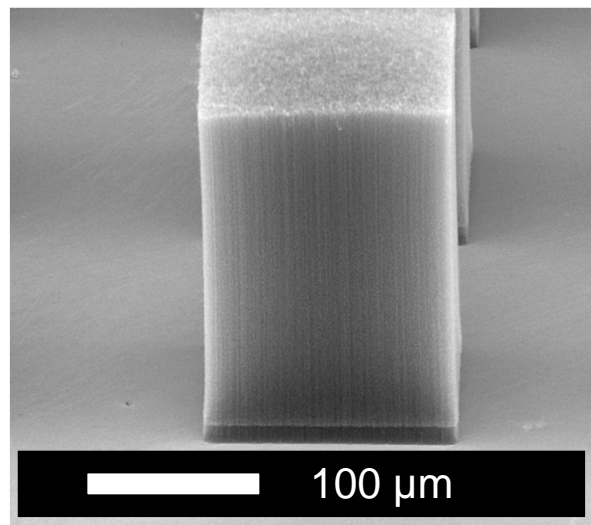
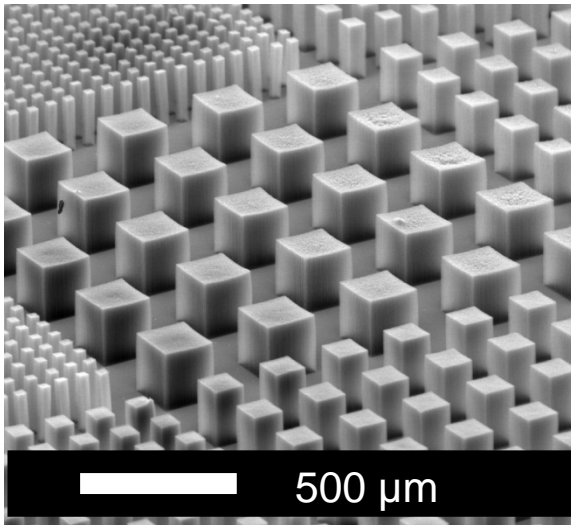
Sunny Wicks, Dr. Roberto Guzman deVilloria & Prof. Brian Wardle
Massachusetts Institute of Technology

Advanced Composites & CNTs



Carbon Nano-Tube (CNT) laminates are a natural progression for aerospace composites due to their superior specific strength & stiffness

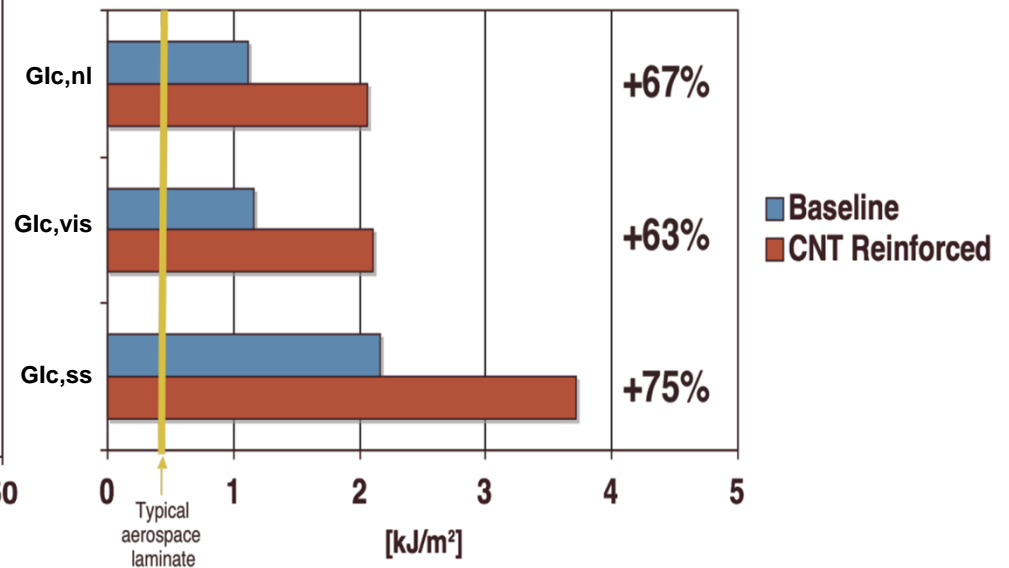
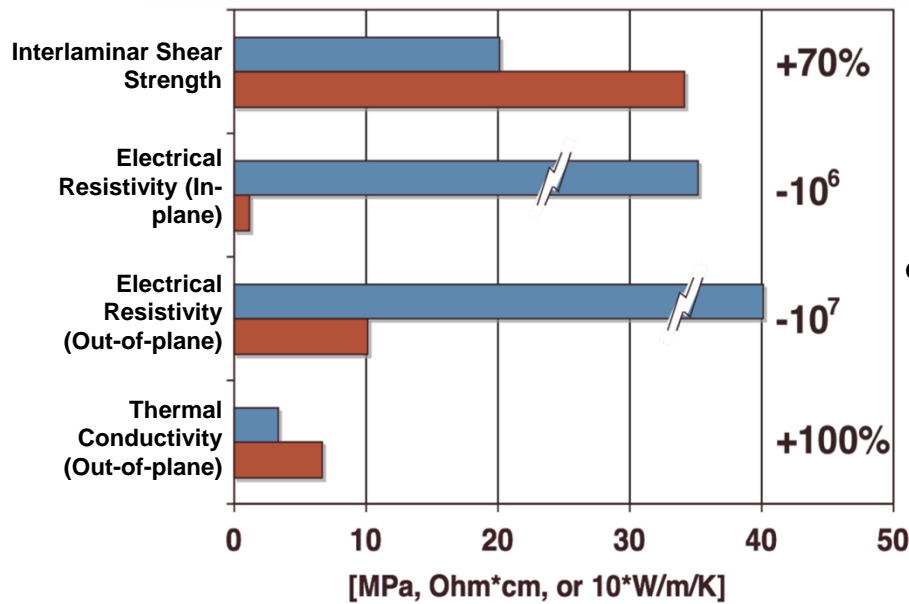
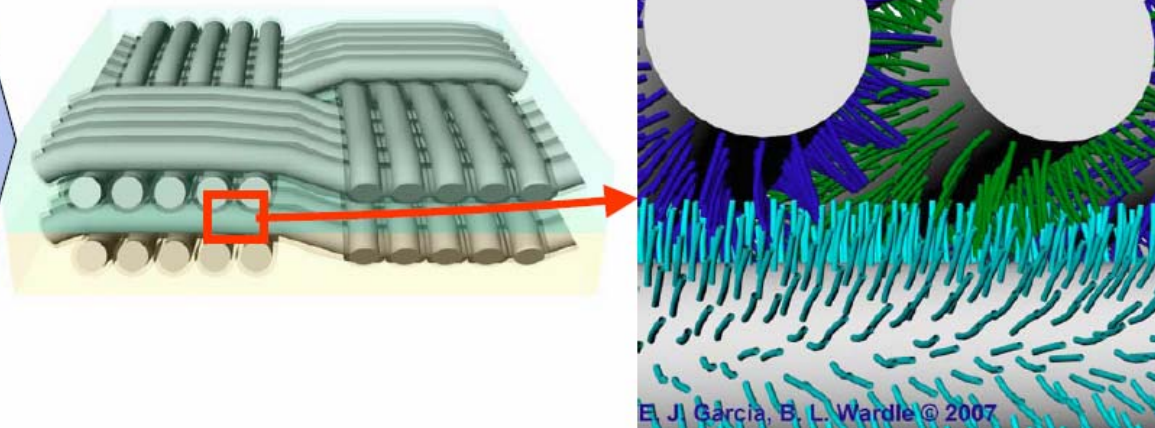
Fabrication of Structured CNTs



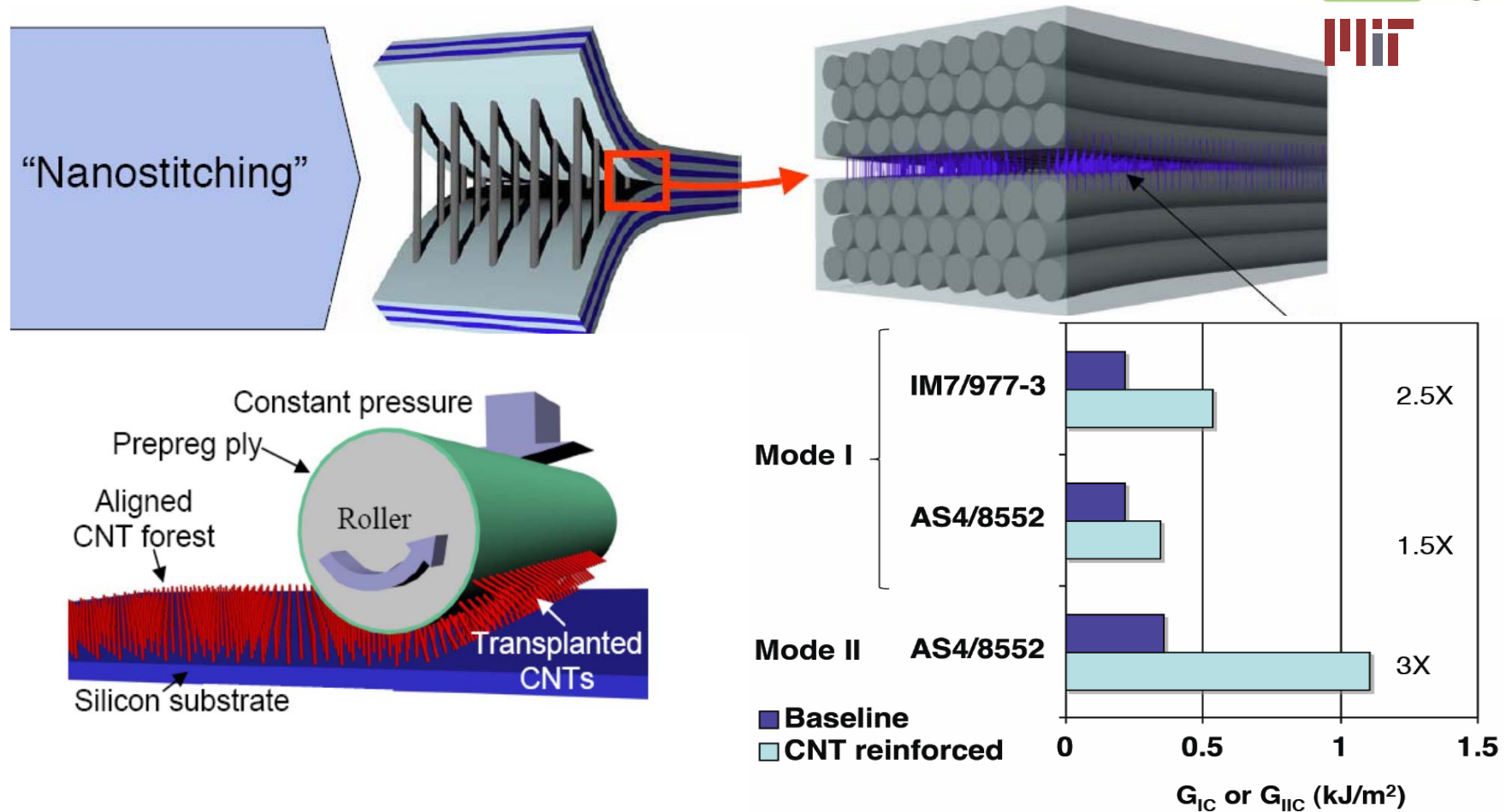
- MIT patented novel CNT fabrication processes
 - CNTs grown aligned directly on fibers or on substrate to be transferred
 - good alignment, dispersion, adhesion & yields high CNT volume fraction
- Atmospheric pressure chemical vapor deposition (CVD)
 - self-aligned morphology 10^{10} - $10^{11}/\text{cm}^2$ of continuous CNTs (7-10 nm OD)
 - rapid forest growth of > 2 microns/second (up to 5 mm long)

"Fuzzy Fiber" Laminates

"Fuzzy Fiber"
Reinforced
Plastic (FFRP)



"Nano-Stitched" Prepreg

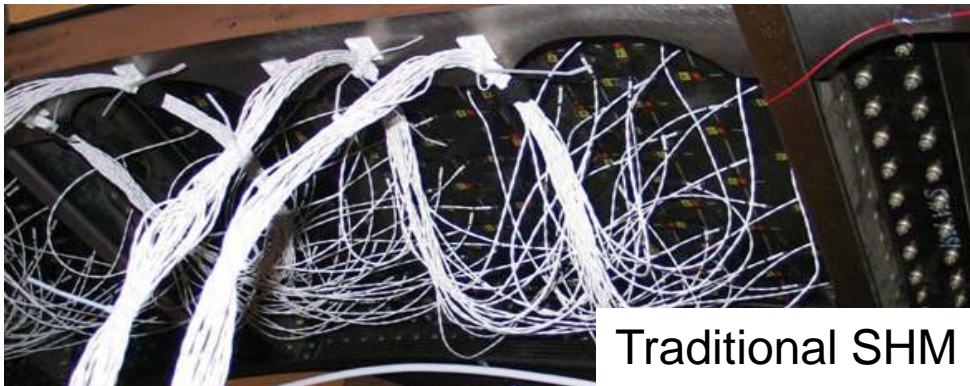


- Grow aligned CNTs on high-temperature substrate
- Transplant CNTs to composite at low temperature
- Process the enhanced composite normally

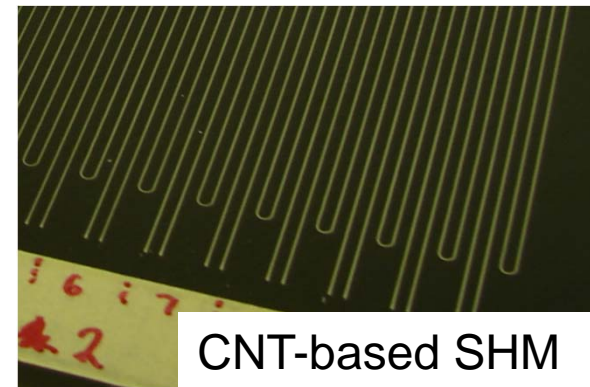
CNT-based SHM

Motivation For CNT-based SHM

- SHM improves reliability, safety & readiness @ reduced costs
 - sensors & cables add weight as well as durability & EMI concerns
 - scaling SHM for large-area coverage has presented challenges
- Advantages of proposed CNT-based SHM methodology
 - sensing elements actually improve specific strength/stiffness of structure
 - damage alters CNT links around affected zone, impacts resistivity
 - surface & sub-surface damage images produced in post-processing
 - simple to scale over large structure, maintains good local resolution

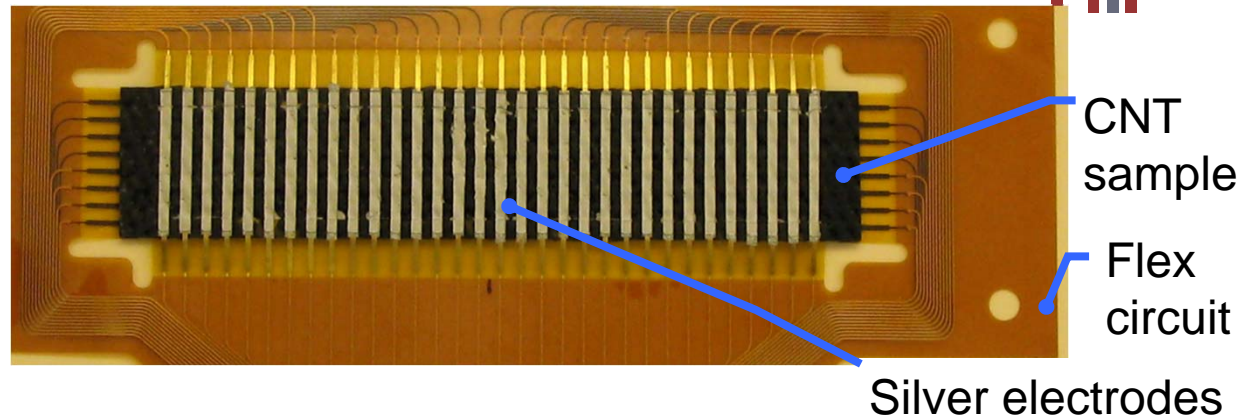
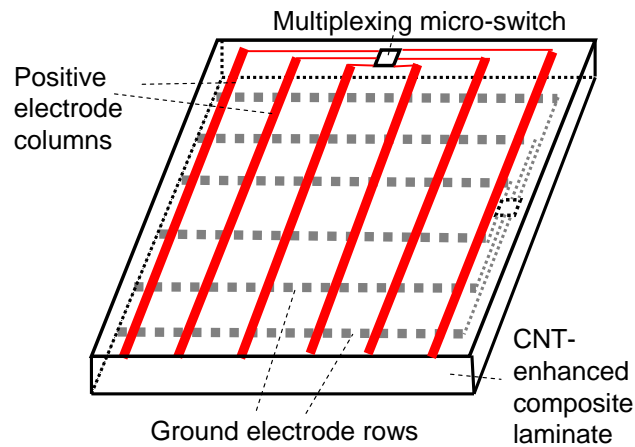


Traditional SHM



CNT-based SHM

SHM Experimental Setup



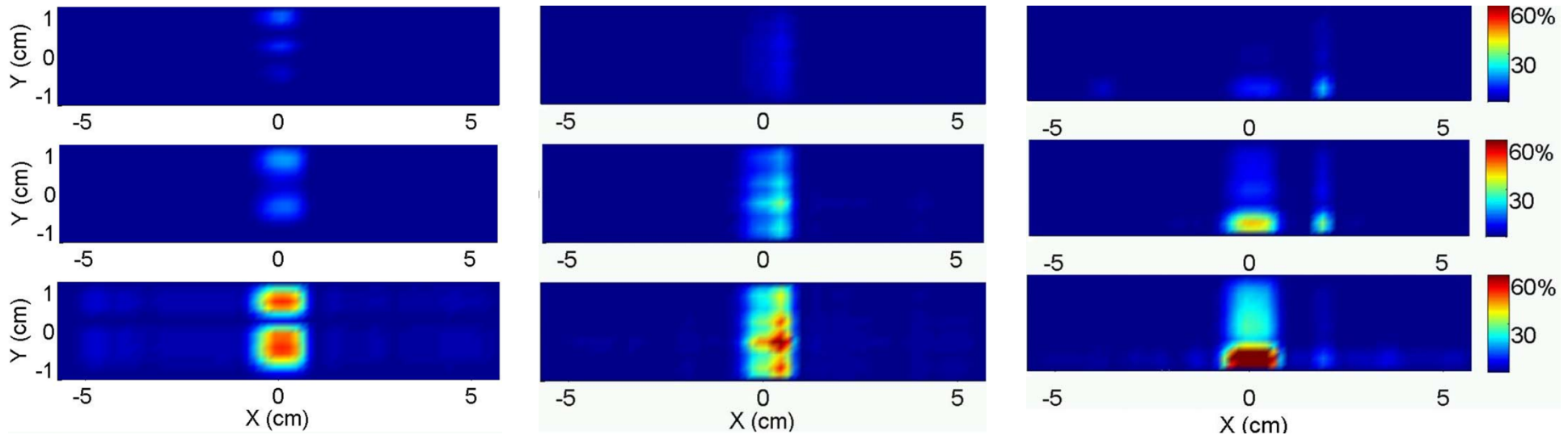
- **FFRP laminates fabricated**

- alumina fiber satin-weave cloth dipped in solution of 50 mM iron nitrate
- CNTs grow radially aligned 20-50 μm with modified thermal CVD method
- 2 plies stacked by hand layup, infused with RTM-6 for 12-hour RT cure
- ~50% fiber volume fraction & ~ 2% CNT (115 x 25 x 2 mm)

- **Silver-ink electrodes applied w/masked silk-screening process**

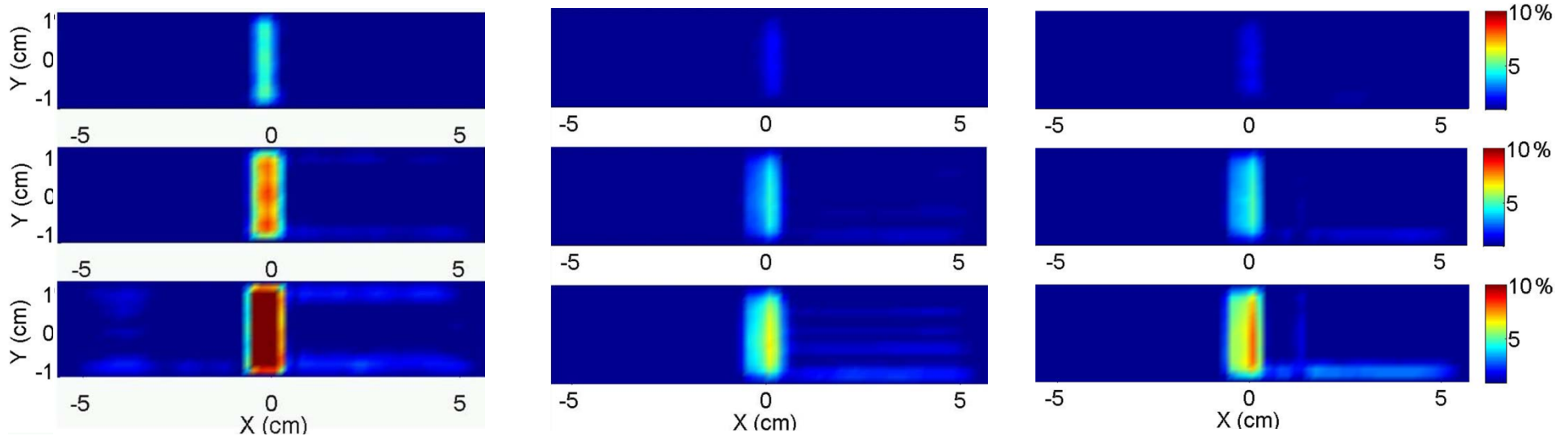
- direct-write (DW) electrode grids applied similar to LCD technology
- 8 x 32 traces 1.5 mm wide, all traces spaced by 1.5 mm
- in-plane & through-thickness resistance measurements collected

In-plane Resistance Changes



- No visible damage was present in any of these cases
 - nearly linear increase in % resistance change with impact energy
 - < 1% change in resistance away from impact zone
- Appeared relatively localized to the actual impacted region
 - 15 ft-lbs impact caused ~10-20% changes
 - 30 ft-lbs impact caused ~20-30% changes
 - 45 ft-lbs impact caused ~40-60% changes

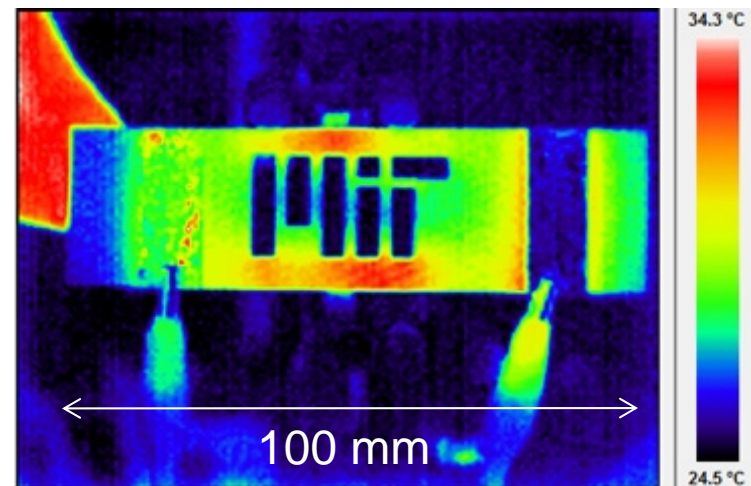
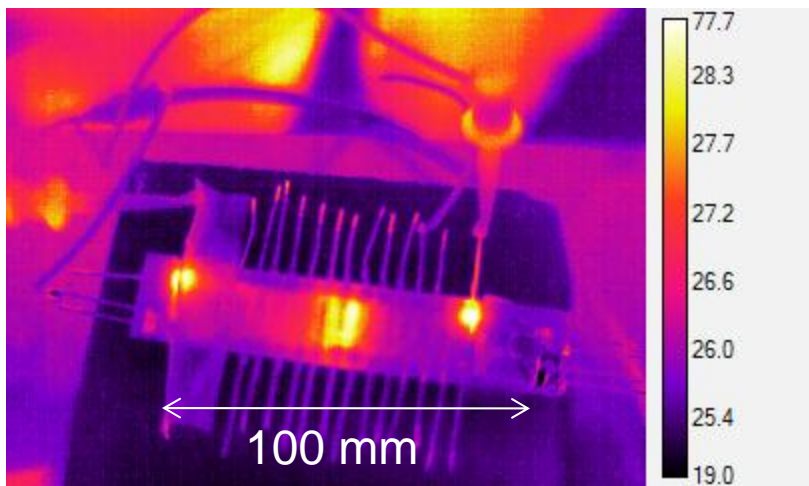
Through-Thickness Changes



- Same trends observed for in-plane vs through-thickness results
 - witness specimen testing indicated complete fracture at 50-60 ft-lbs
 - no visible micro-cracking until failure
- Appeared to effect width in impact region relatively uniformly
 - 15 ft-lbs impact caused ~2-4% changes
 - 30 ft-lbs impact caused ~4-8% changes
 - 45 ft-lbs impact caused ~8-10% changes

NDE Approaches

- Schemes being explored for NDE & Quality Control
 - acoustic emission (AE) measuring dynamic piezoresistive changes
 - enhanced penetrating thermographic NDE with applied voltage
- Same hardware & flex frame can be used to measure dynamic resistance changes or self-induce heating
 - initial “pencil-tap” experiments verifies that AE can be detected
 - initial thermographs demonstrate method feasibility



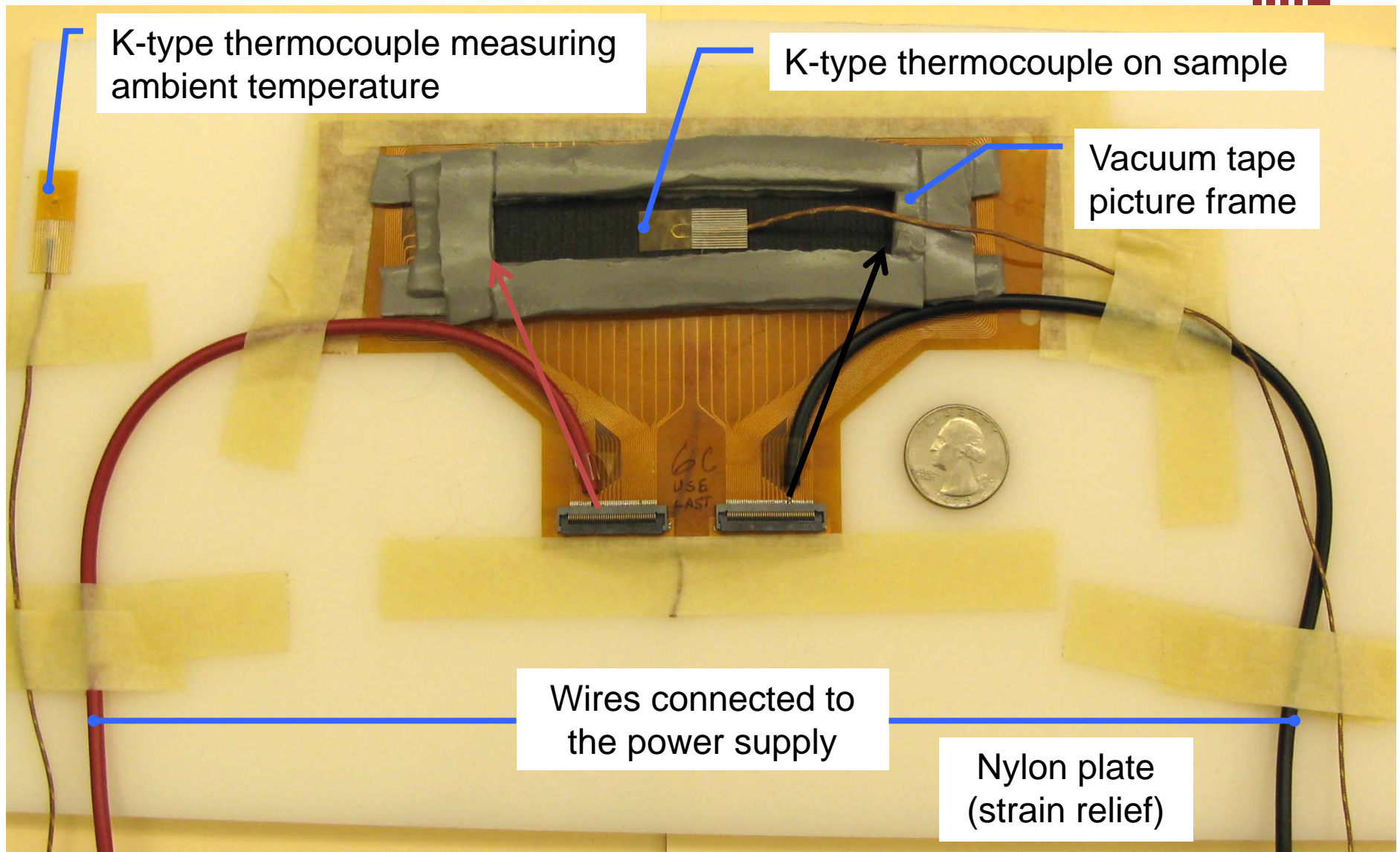
CNT-based IPS

Motivation For CNT-based IPS

- Goal to develop a multi-role system for composite aerosurfaces
 - detection of presence of ice on surface
 - removal of ice and/or prevention of ice (re)formation
 - detection/characterization of structural damage
- Current approaches provides high false positive & failure rates
- **CNT-based IPS to produce robust, reliable integrated solution**
 - heating/sensing elements are structural
 - solid state (no moving parts)
 - conformal (light & low profile)
 - uniform surface coverage
 - efficient closed-loop feedback possible
 - can improve impact resistance



Proof-of-Concept Experiments



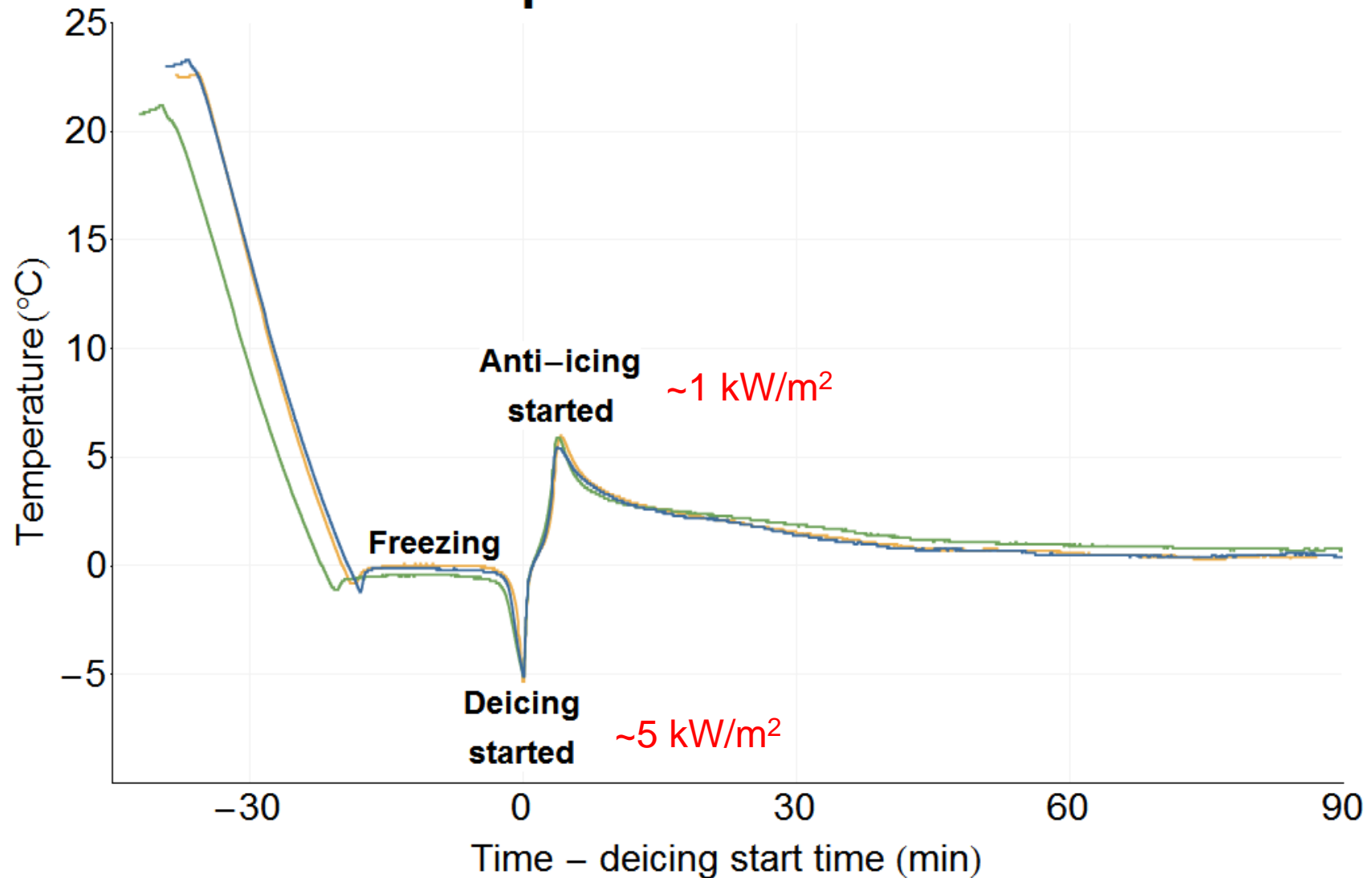
Proof-of-Concept Ice Protection



- Anti-Icing @ 4 W input ($\sim 2 \text{ kW/m}^2$)
 - without water $\sim 30^\circ\text{C}$ difference in CNT asymptotic temperature
 - able to maintain sample temperature above freezing with large margin
 - visual confirmation that ice did not form
- De-Icing @ 4 W input ($\sim 2 \text{ kW/m}^2$)
 - asymptotic values only a function of power regardless of water/ice
 - time to asymptote only a function of water/ice regardless of power level
 - higher power level provide steeper slope through 0°C (de-ice quicker)
- Ice-Detection @ 4 W input ($\sim 2 \text{ kW/m}^2$)
 - while heating with water, distinct phase-related slopes
 - effective heat capacity of melting ice
 - uses anti/de-icing setup without any additional electrodes

De-Icing & Anti-Icing

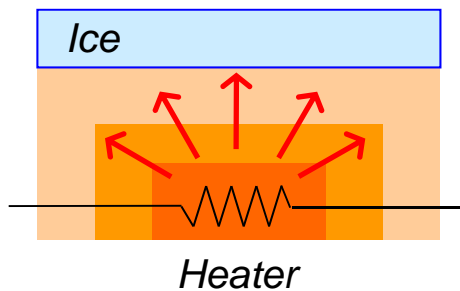
Temperature versus time



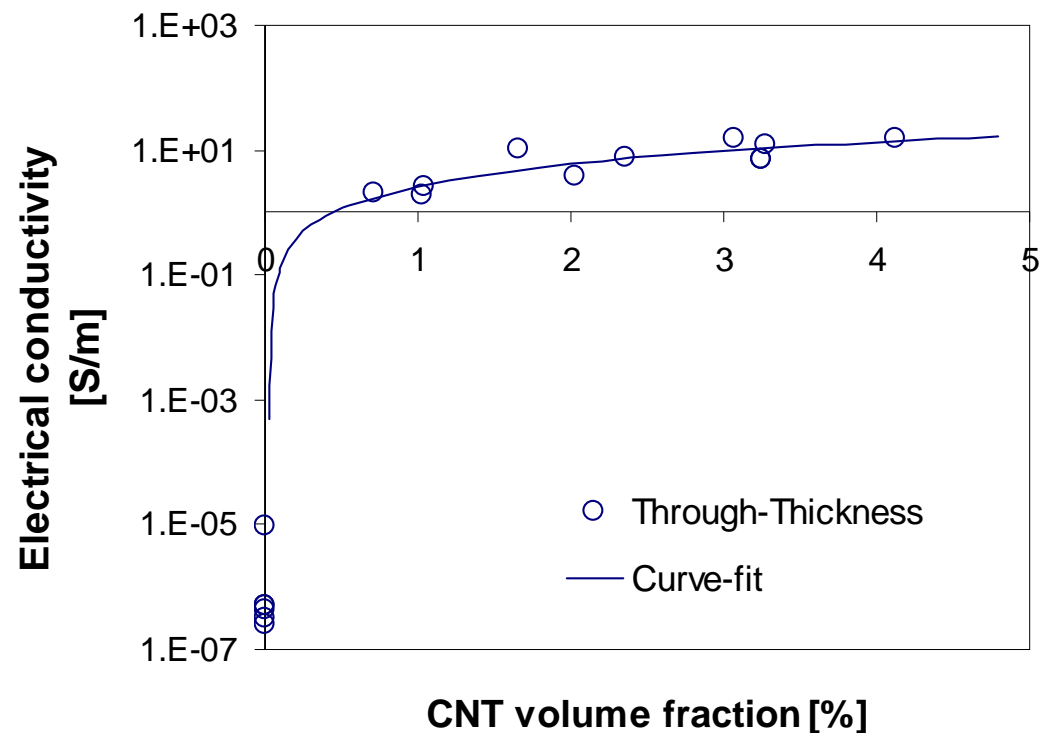
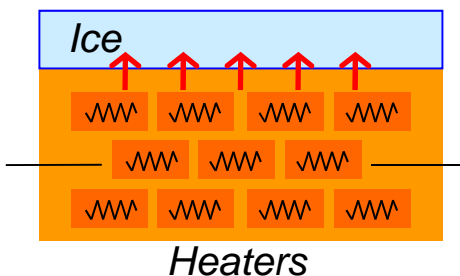
Benefits of CNT Deicing Layer

- Quasi-uniform resistive heating across large areas
- Effective heat distribution for large areas
- Tunable resistivity (material property) for optimal power setting

Current deicing design



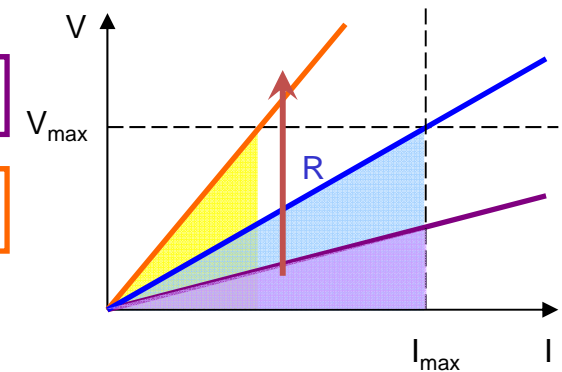
CNT deicing layer



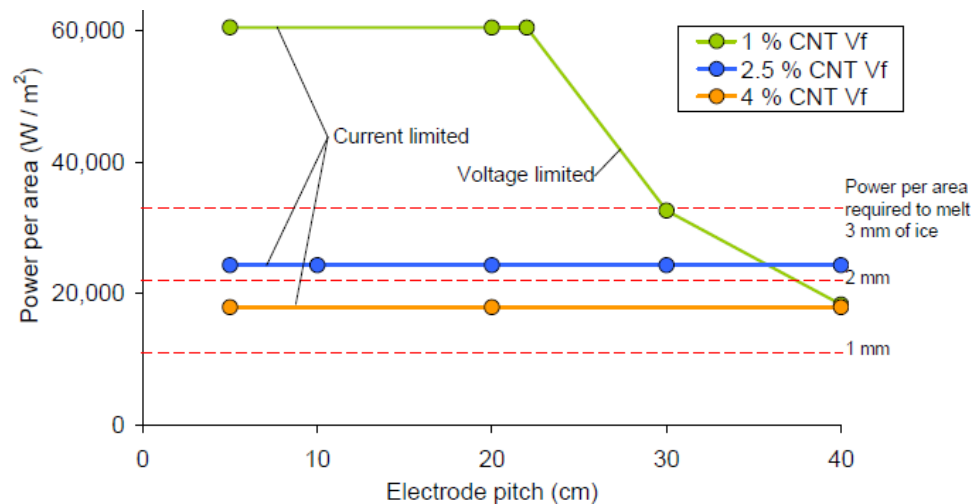
Optimal Resistance for Max Power

If R small, current limited, and thus $\text{Power} = R I_{\max}^2$ (linear)

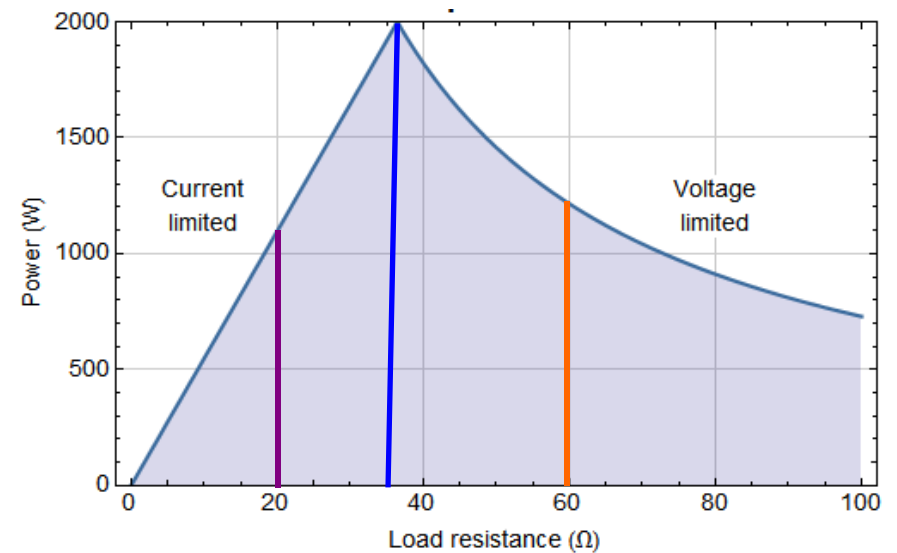
If R large, voltage limited, and thus $\text{Power} = V_{\max}^2 / R$



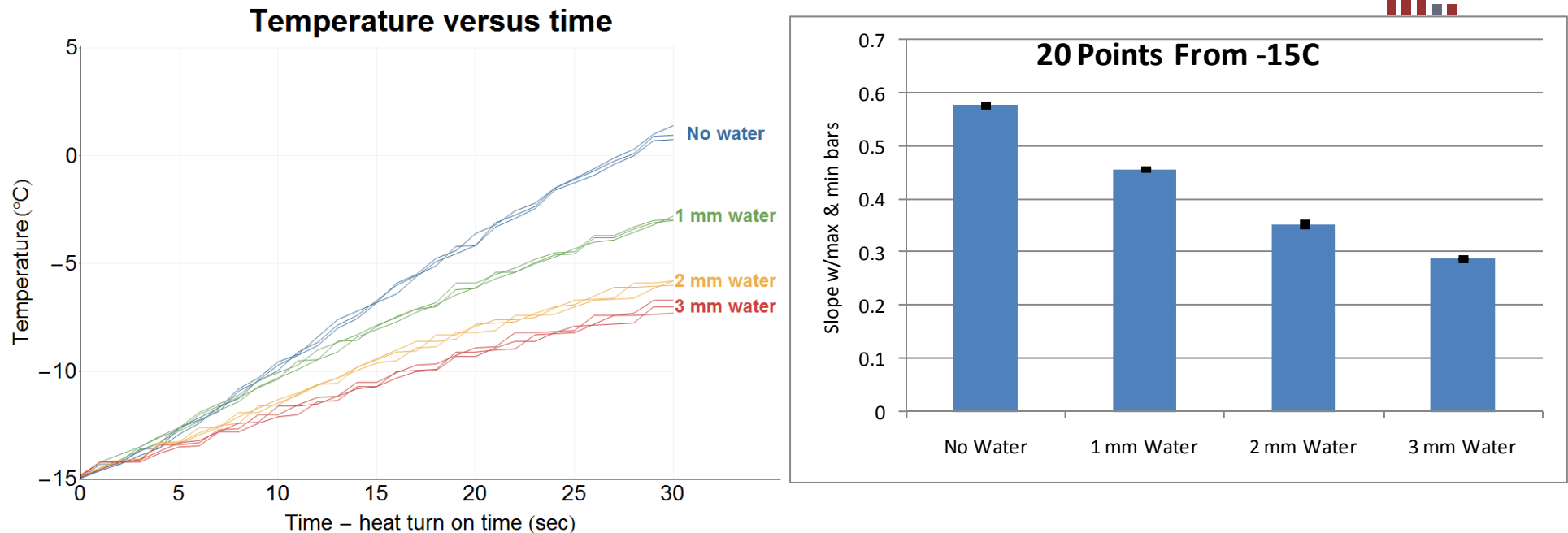
Power/Area with Varying Electrode Spacing



Dissipated Power versus Load Resistance



Ice-Detection Formulation



- Effective heat capacity method
 - constant current applied to sample for fixed temperature recording time
 - exponential temperature rise (linear data fit), compared to no-ice case
- Very repeatable results
 - shallower slope correlates to more ice on specimen
 - data consistent for multiple detection temperatures

Summary



- Proof-of-concepts for CNT-based SHM & IPS demonstrated
 - Reliable – solid state, structural elements, durability & longevity
 - Minimal impact – low mass & low profile
 - Integrated solution – ice-detection & de-icing + damage detection & NDE
- CNT-based SHM
 - LCD inspired design for in-plane and through-thickness detection
 - damage affects CNT-links, can measure resistivity changes
 - resolution defined by grid spacing; easily modified/expanded
- CNT-based IPS
 - anti-icing @ $\sim 1 \text{ kW/m}^2$ to maintain blade temperature $\sim 5^\circ \text{C}$
 - de-icing @ $\sim 5 \text{ kW/m}^2$, ~ 2 min from $\sim -15^\circ \text{C}$ (not including gravity assist)
 - ice-detection in seconds (faster with faster/finer DAQ)

Continuing Research



- **CNT-based SHM system (AFOSR Phase II funding)**
 - measure structural & multi-functional properties for CFRP FFRP
 - explore various electroding strategies
 - analytical models to simulate system, parametric iteration
 - find trade between detection resolution & electrode spacing
 - compensation algorithms for temperature & loading
 - demonstrate on UAV wing/tail section
- **CNT-based IPS system (NAVAIR Phase II funding)**
 - determine electrode spacing versus heating efficiency
 - design of laminate morphology for surface heating
 - development of hardware for deicing and ice detection
 - fabrication & demonstration of BAMS leading edge IPS
 - ice-tunnel testing

Acknowledgments



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 - AFOSR FA9550-09-C-0165 “SHM of CNT-Enhanced Composites”
 - NAVAIR N68335-10-C-0227 “Composite Self-Monitoring Anti/De-icing”
 - NASA NNX09CC57P “Cable-Free Sensor-Bus for SHM of Composites”
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 - Nano-Engineered Composite aerospace STructures consortium
 - supported by large aerospace OEMs & composite fabricators
 - Airbus, Boeing, Embraer, Lockheed, Saab, Spirit AeroSystems, Textron, Composite Systems Technology & TohoTenax

